

AEROSPACE MATERIAL SPECIFICATION

SAE,

AMS-QQ-A-250/5A

Issued Revised Reaffirmed 1997-08 1998-09 2010-05

Superseding AMS-QQ-A-250/5

Aluminum Alloy Alclad 2024, Plate and Sheet

A82024

RATIONALE

This document has been reaffirmed to comply with the SAE 5-year Review policy

NOTICE

This document has been taken directly from Federal Specification QQ-A-250/5F, Amendment 2, and contains only minor editorial and format changes required to bring it into conformance with the publishing requirements of SAE technical standards.

The original Federal Specification was adopted as an SAE standard under the provisions of the SAE Technical Standards Board (TSB) Rules and Regulations (TSB 001) pertaining to accelerated adoption of government specifications and standards. TSB rules provide for (a) the publication of portions of unrevised government specifications and standards without consensus voting at the SAE Committee level, (b) the use of the existing government specification or standard format, and (c) the exclusion of any qualified product list (QPL) sections.

The complete requirements for procuring 2024 aluminum alloy alclad plate and sheet described herein shall consist of this document and the latest issue of AMS-QQ-A-250.

SAE Technical Standards Board Rules provide that: "This report is published by SAE to advance the state of technical and engineering sciences. The use of this report is entirely voluntary, and its applicability and suitability for any particular use, including any patent infringement arising therefrom, is the sole responsibility of the user."

SAE reviews each technical report at least every five years at which time it may be reaffirmed, revised, or cancelled. SAE invites your written comments and suggestions. Copyright © 2010 SAE International

All rights reserved. No part of this publication may be reproduced, stored in a retrieval system or transmitted, in any form or by any means, electronic, mechanical, photocopying, recording, or otherwise, without the prior written permission of SAE.

TO PLACE A DOCUMENT ORDER:

SAE WEB ADDRESS:

Tel: 877-606-7323 (inside USA and Canada) Tel: +1 724-776-4970 (outside USA)

Fax: 724-776-0790

Email: CustomerService@sae.org

http://www.sae.org

SAE values your input. To provide feedback on this Technical Report, please visit

http://www.sae.org/technical/standards/AMSQQA250/5A

- 1. SCOPE AND CLASSIFICATION:
- 1.1 Scope:

This specification covers the specific requirements for 2024 aluminum alloy alclad plate and sheet; the general requirements are covered in AMS-QQ-A-250. The plate and sheet covered by this specification shall be an integral composite product consisting of a heat-treatable aluminum alloy 2024 core with thin layers of an aluminum alloy 1230 (99.30 percent minimum aluminum) anodic to the core and of approximately equal thickness, bonded to both surfaces.

- Classification: 1.2
- Tempers: The plate and sheet are classified in one of the following tempers as specified (See 6.2 1.2.1 ien the full Pof of a and 6.4): O, T3, T4, T42, T62, T72, T81, T351, T361, T851, T861, or F temper. Definitions of these tempers are specified in AMS-QQ-A-250.
- 2. APPLICABLE DOCUMENTS:

See AMS-QQ-A-250.

- 3. REQUIREMENTS:
- 3.1 Chemical Composition:
- The chemical composition of the core ingots or slabs and of the cladding plates used for the 3.1.1 manufacture of the alclad plate and sheet shall conform to the requirements specified in Table I for core and cladding, respectively.

TABLE I. Chemical Composition 1/

| Element | (| Claddng 1230 | | |
|----------------------------------|----------------|-----------------|-----------------|--|
| | Minimum | Maximum | Maximum | |
| | <u>Percent</u> | <u>Percent</u> | <u>Percent</u> | |
| Copper | 3.8 | 4.9 | 0.10 | |
| Magnesium | 1.2 | 1.8 | 0.05 | |
| Manganese | 0.30 | 0.9 | 0.05 | |
| Iron | - | 0.50 | 2/ | |
| Silicon | - | 0.50 | <u>2</u> / | |
| Chromium | - | 0.10 | - | |
| Nickel | - | - | _ | |
| Zinc | - | 0.25 | 0.10 | |
| Titanium | - | 0.15 | 0.03 | |
| Other Elements, each | - | 0.05 | 0.03 <u>3</u> / | |
| Other Elements, total <u>4</u> / | - | 0(15 | | |
| Aluminum | - | Remainder | 99.30 Minimum | |

- 1/ Analysis shall routinely be made only for the elements specifically mentioned in Table I. If, however, the presence of other elements is indicated or suspected in amounts greater than the specified limits, further analysis shall be made to determine that these elements are not present in excess of specified limits.
- 2/ Iron plus silicon, 0.7 percent, maximum.
- 3/ Vanadium 0.05 percent maximum.
- 4/ The sum of those "Other" metallic elements 0.010 percent or more each, expressed to the second decimal before determining the sum.
- 3.2 Mechanical Properties:
- 3.2.1 Mechanical Properties of Material as Supplied: The mechanical properties perpendicular to the direction of final rolling, except for material under 9 inches in width, shall conform to the requirements of Table II for the temper specified. For material under 9 inches in width, the mechanical properties parallel to the direction of final rolling shall conform to the requirements of Table II for the temper specified.

TABLE II. Mechanical Properties (See 6.6)

| Temper | Widths | Thickness | Tensile Strength minimum | Yield Strength at 0.2 percent Offset or at Extension Indicated | | Elongation in 2 in. or 4 times D <u>1</u> /, <u>2</u> /, minimum |
|-----------------|--------|-----------------------------|--------------------------------|---|--------------------------------|---|
| | Inches | Inches | ksi | Minimum ksi | Extension under load Inch/Inch | Percent |
| 0 | All | 0.008 thru 0.009 | 30.0 <u>3</u> / | 14.0 <u>3</u> / | 0.0035 | 10 |
| | All | 0.010 thru 0.062 | 30.0 <u>3</u> / | 14.0 <u>3</u> / | 0.0035 | 12 |
| | All | 0.063 thru 0.499 | 32.0 <u>3</u> / | 14.0 3 | 0.0036 | 12 |
| | All | 0.500 thru 1.750 <u>4</u> / | 32.0 <u>3</u> / | \$.0 | | 12 |
| T3 <u>5</u> / | All | 0.008 thru 0.009 | 58.0 | 39.0 | 0.0061 | 10 |
| (See 6.4) | All | 0.010 thru 0.020 | 59.0 | 39.0 | 0.0061 | 12 |
| | All | 0.021 thru 0.062 | 59.0 | 39.0 | 0.0061 | 15 |
| | All | 0.063 thru 0.128 | 61,0 | 40.0 | 0.0060 | 15 |
| | All | 0.129 thru 0.249 | 62.0 | 40.0 | 0.0060 | 15 |
| T4 <u>6</u> / | All | 0.010 thru 0.020 | 58.0 | 36.0 | 0.0059 | 12 |
| | All | 0.021 thru 0.062 | 58.0 | 36.0 | 0.0059 | 15 |
| | All | 0.063 thru 0.128 | 61.0 | 38.0 | 0.0058 | 15 |
| T361 <u>7</u> / | All | 0.020 thru 0.062 | 61.0 | 47.0 | 0.0070 | 8 |
| | All | 0.063 thru 0.499 | 64.0 | 48.0 | 0.0071 | 9 |
| | All | 0.500° <u>4</u> / | 66.0 | 49.0 | 0.0071 | 10 |
| T42 <u>8</u> / | All | 0008 thru 0.009 | 55.0 | 34.0 | 0.0056 | 10 |
| | All . | 0.010 thru 0.020 | 57.0 | 34.0 | 0.0056 | 12 |
| | All | 0.021 thru 0.062 | 57.0 | 34.0 | 0.0056 | 15 |
| | AID. | 0.063 thru 0.249 | 60.0 | 36.0 | 0.0056 | 15 |
| | All | 0.250 thru 0.499 | 60.0 | 36.0 | 0.0058 | 12 |
| | SAII | 0.500 thru 1.000 <u>4</u> / | 61.0 | 38.0 | 0.0055 | 8 |
| | All | 1.001 thru 1.500 <u>4</u> / | 60.0 | 38.0 | 0.0056 | 7 |
| | All | 1.501 thru 2.000 <u>4</u> / | 60.0 | 38.0 | 0.0055 | 6 |
| | All | 2.001 thru 3.000 <u>4</u> / | 58.0 | 38.0 | 0.0055 | 4 |
| T351 | All | 0.250 thru 0.499 | 62.0 | 40.0 | 0.0060 | 12 |
| | All | 0.500 thru 1.000 <u>4</u> / | 63.0 | 42.0 | 0.0060 | 8 |
| | All | 1.001 thru 1.500 <u>4</u> / | 62.0 | 42.0 | 0.0060 | 7 |
| | All | 1.501 thru 2.000 <u>4</u> / | 62.0 | 42.0 | 0.0060 | 6 |
| | All | 2.001 thru 3.000 <u>4</u> / | 60.0 | 42.0 | 0.0060 | 4 |
| | All | 3.001 thru 4.000 <u>4</u> / | 57.0 | 41.0 | 0.0058 | 4 |

TABLE II. Mechanical Properties (See 6.6) (Continued)

| | | | | • | | |
|-----------------|--------|--------------------|--------------------------------|---------------------|--|---|
| Temper | Widths | Thickness | Tensile Strength minimum | 0.2 perc or at E | rength at ent Offset xtension cated | Elongation in 2 in. or 4 times D <u>1</u> /, <u>2</u> /, minimum |
| | | | | | Extension | |
| | | | | Minimum | under load | ^ |
| | Inches | Inches | ksi | ksi | Inch/Inch | Percent |
| T62 <u>8</u> / | All | 0.010 thru 0.062 | 60.0 | 47.0 | 0.0070 / | 5 |
| | All | 0.063 thru 0.499 | 62.0 | 49.0 | 0.0071 | 5 |
| T72 <u>8</u> / | All | 0.010 thru 0.062 | 56.0 | 43.0 | 0.0066 | 5 |
| | All | 0.063 thru 0.249 | 58.0 | 45.0 | 0.0068 | 5 |
| T81 <u>5</u> / | All | 0.010 thru 0.062 | 62.0 | 54.00 | 0.0077 | 5 |
| | All | 0.063 thru 0.249 | 65.0 | 56.0 | 0.0079 | 5 |
| T861 <u>7</u> / | All | 0.020 thru 0.062 | 64.0 | 58.0 | 0.0081 | 3 |
| | All | 0.063 thru 0.249 | 69.0 | 64.0 | 0.0084 | 4 |
| | All | 0.250 thru 0.499 | 68.0 | 62.0 | 0.0082 | 4 |
| | All | 0.500 <u>4</u> / | 70.0 | 64.0 | 0.0081 | 4 |
| T851 | All | 0.250 thru 0.499 | 65.0 | 56.0 | 0.0075 | 5 |
| | All | 0.500 thru 1.000 4 | 66.0 | 58.0 | 0.0075 | 5 |
| F | All | All . | <u>9</u> / | <u>9</u> / | <u>9</u> / | <u>9</u> / |

- 1/ Not required for material 1/2 inch or less in width.
- 2/ D represents specimen diameter.
- 3/ Maximum.
- 4/ These properties are those of the core alloy since the tests are made on a round specimen machined from the plate.
- 5/ Applicable to flat sheet only.
- 6/ Applicable to coiled sheet only.
- 7/ Applicable to flat sheet and plate only.
- 8/ Material in the 742, T62, or T72 temper is not available from material producers.
- 9/ No requirements.
- 3.2.2 Mechanical Properties After Heat Treatment: In addition to conforming to the requirements of 3.2.1, material in the annealed (O) and the as-fabricated (F) tempers shall, after proper solution heat-treatment, also conform to the requirements of Table II for the T42 temper. Material as received in the T3, T4, T351, T81, and T851 tempers shall, after proper re-solution heat-treatment, be capable of conforming to the requirements specified in Table II for the T42 temper. Material in the T42 temper shall, after proper aging, be capable of conforming to the requirements specified in Table II for the T62 or T72 temper. Material in the T3, T351, and T361 tempers shall, after proper aging, be capable of conforming to the requirements specified in Table II for the T81, T851, and T861 tempers, respectively.

3.2.3 Bend Test: Bend specimens taken from material shall be capable of withstanding, without cracking, the bend test specified in AMS-QQ-A-250. The values for bend factor N are given in Table III.

TABLE III. Bend Test Factor "N"

| | Tempers | | | • | |
|--------------------------|---------|------|------------------|------------|-----------------|
| Thickness of Material | 0 | Т3 | T4 and T42 | T361 | \$\frac{1}{2}\$ |
| Inch | | | | | Jy, |
| 0.008 thru 0.009 | 0 | 4 | 4 | - | 70,0 |
| 0.010 thru 0.032 | 0 | 4 | 4 | 4 💉 | S |
| 0.033 thru 0.040 | 1 | 4 | 4 | 40 | |
| 0.041 thru 0.062 | 1 | 5 | 5 | <u></u> 04 | |
| 0.063 thru 0.128 | 2 | 5 | 5 | 6 | |
| 0.129 thru 0.187 | 2 | 8 | 8 | 6 | |
| 0.188 thru 0.249 | 2 | 8 | 8 | 8 | |
| 0.250 thru 0.499 | 2 | Ve . | 10 | - | |

- 3.3 Cladding Thickness:
- 3.3.1 Thickness of Cladding Plates: The aluminum alloy plates that are bonded to the two sides of the aluminum alloy (2024) ingot or slab, to form a composite that is to be rolled, shall each have a thickness as specified in Table IV.

TABLE IV. Cladding Thickness

| Thickness of finished plate or sheet | Nominal cladding thickness per side, percent of composite thickness | Average thickness per side of cladding on finished plate or sheet, minimum Percent of plate or sheet thickness |
|--------------------------------------|---|---|
| Inches | | |
| Under 0.063 | 5 | 4 |
| 0.063 and over | 2-1/2 | 2 |

3.3.2 Thickness of Cladding: If question arises concerning the thickness of cladding of the finished sheet or plate, samples examined in accordance with AMS-QQ-A-250 shall show an averge thickness of cladding on each side, not less than that specified in Table IV.