AERONAUTICAL MATERIAL SPECIFICATION

Society of Automotive Engineers, Inc. 29 West 39th Street New York City AMS4725 A

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COPPER-BERYLLIUM ALLOY WIRE 98Cu - 1.9Be Solution Treated

- 1. ACKNOWLEDGMENT: A vendor shall mention this specification number and its revision letter in all quotations and when acknowledging purchase orders.
- 2. APPLICATION: Primarily for parts requiring high strength with good electrical conductivity or lack of magnetic susceptibility.
- 3. COMPOSITION:

Beryllium
Nickel + C

1.8 ~2.09

Nickel + Cobalt

0,20 min

Nickel + Cobalt + Iron

0.60 max

Copper + Total Named Elements

499.5 min

- 4. CONDITION: Cold drawn or rolled and solution heat treated, in a suitable condition for precipitation heat treatment.
- 5. TECHNICAL REQUIREMENTS:
- 5.1 Tensile Properties:

Tensile Strength, psi Elongation, % in 2 in.

58,000 - 78,000 35 min

5.2 Tensile Properties After Precipitation Heat Treatment: Wire after being heated at 600 F + 5 for 3 hr and cooling in air shall conform to the following requirements:

Tensile Strength, psi Elongation, % in 2 in.

160,000 min 1.5 min

- 6. QUALITY: Wire shall be uniform in quality and condition, clean, sound, smooth, and free from foreign materials and internal and external defects detrimental to fabrication or to performance of parts.
- 7. TOLERANCES: Unless otherwise specified, tolerances shall conform to the latest issue of AMS 2224 as applicable. Diameter, thickness and width tolerances shall be as specified below:
- 7.1 Rounds, Hexagons and Octagons: Table II, Refractory.
- 7.2 Squares: Table III, Refractory.
- 7.3 Rectangles, Thickness: Table III, Refractory.
- 7.4 Rectangles, Width: Table IV, Refractory.