

AEROSPACE MATERIAL SPECIFICATION

SAE AMS4211

REV. B

Issued 2000-07
Revised 2007-04
Reaffirmed 2012-09

Superseding AMS4211A

Aluminum Alloy, Plate
6.2Zn - 1.9Cu - 2.1Mg - 0.10Zr (7040-T7451)
Solution Heat Treated, Stress Relieved, and Overaged
(Composition similar to UNS A97040)

RATIONALE

AMS4211B has been reaffirmed to comply with the SAE five-year review policy.

1. SCOPE

1.1 Form

This specification covers an aluminum alloy in the form of plate.

1.2 Application

This plate has been used typically for parts requiring a high level of mechanical properties and good resistance to stress-corrosion cracking, but usage is not limited to such applications.

2. APPLICABLE DOCUMENTS

The issue of the following documents in effect on the date of the purchase order forms a part of this specification to the extent specified herein. The supplier may work to a subsequent revision of a document unless a specific document issue is specified. When the referenced document has been cancelled and no superseding document has been specified, the last published issue of that document shall apply.

2.1 SAE Publications

Available from SAE International, 400 Commonwealth Drive, Warrendale, PA 15096-0001, Tel: 877-606-7323 (inside USA and Canada) or 724-776-4970 (outside USA), or www.sae.org.

AMS 2355	Quality Assurance Sampling and Testing of Aluminum Alloys and Magnesium Alloys, Wrought Products (Except Forging Stock) and Flash Welded Rings
AMS 2772	Heat Treatment of Aluminum Alloy Raw Materials
AS1990	Aluminum Alloy Tempers

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2.2 ASTM Publications

Available from ASTM International, 100 Barr Harbor Drive, P.O. Box C700, West Conshohocken, PA 19428-2959, Tel: 610-832-9585, or www.astm.org.

ASTM B 594	Ultrasonic Inspection of Aluminum-Alloy Wrought Products for Aerospace Applications
ASTM B 660	Packing/Packaging of Aluminum and Magnesium Products
ASTM B 666/B 666M	Identification Marking of Aluminum and Magnesium Products
ASTM E 466	Conducting Constant Amplitude Axial Fatigue Tests of Metallic Materials
ASTM G 34	Exfoliation Corrosion Susceptibility in 2xxx and 7xxx Series Aluminum Alloys (EXCO Test)
ASTM G 47	Determining Susceptibility to Stress-Corrosion Cracking of High-Strength Aluminum Alloy Products

2.3 ANSI Publications

Available from American National Standards Institute, 25 West 43rd Street, New York, NY 10036, Tel: 212-642-4900, www.ansi.org.

ANSI H35.2	Dimensional Tolerances for Aluminum Mill Products
ANSI H35.2M	Dimensional Tolerances for Aluminum Mill Products (Metric)

3. TECHNICAL REQUIREMENTS

3.1 Composition

Shall conform to the percentages by weight shown in Table 1, determined in accordance with AMS 2355.

TABLE 1 - COMPOSITION

Element	min	max
Silicon	--	0.10
Iron	--	0.13
Copper	1.5	2.3
Manganese	--	0.04
Magnesium	1.7	2.4
Chromium	--	0.04
Zinc	5.7	6.7
Titanium	--	0.06
Zirconium	0.05	0.12
Other Elements, each	--	0.05
Other Elements, total	--	0.15
Aluminum	remainder	

3.2 Solution Heat Treat at 880 to 900 °F (471 to 482 °C). Stretch to produce a nominal permanent set of 2% but not less than 1-1/2% nor more than 3%, and artificially age to the T7451 temper in accordance with AMS 2772 (See AS1990).

3.2.1 Plate shall receive no further straightening operations after stretching.

3.3 Properties

Plate shall conform to the following requirements, determined in accordance with AMS 2355 on the mill produced product and as specified herein.

3.3.1 Tensile Properties

Shall be as specified in Table 2.

TABLE 2A - MINIMUM TENSILE PROPERTIES, INCH/POUND UNITS

Nominal Thickness Inches	Specimen Orientation	Tensile Strength ksi	Yield Strength at 0.2% Offset ksi	Elongation in 2 inches or 4D %
3.0 to 4.0, incl	Longitudinal	70.0	62.0	9
	Long-Trans.	71.0	62.0	6
	Short Trans.	69.0	59.0	3
Over 4.0 to 5.0, incl	Longitudinal	70.0	62.0	9
	Long-Trans.	71.0	62.0	5
	Short Trans.	68.0	58.0	3
Over 5.0 to 6.0, incl	Longitudinal	70.0	62.0	8
	Long-Trans.	70.0	61.0	4
	Short Trans.	68.0	58.0	3
Over 6.0 to 7.0, incl	Longitudinal	69.0	62.0	7
	Long-Trans.	69.0	60.0	4
	Short Trans.	67.0	57.0	3
Over 7.0 to 8.0, incl	Longitudinal	68.0	61.0	6
	Long-Trans.	69.0	60.0	4
	Short Trans.	66.0	57.0	3
Over 8.0 to 9.0, incl	Longitudinal	68.0	61.0	6
	Long-Trans.	68.0	59.0	4
	Short Trans.	66.0	56.0	3

TABLE 2B - MINIMUM TENSILE PROPERTIES, SI UNITS

Nominal Thickness Millimeters	Specimen Orientation	Tensile Strength MPa	Yield Strength at 0.2% Offset MPa	Elongation in 50.8 mm or 5D %
76 to 102, incl	Longitudinal	483	427	8
	Long-Trans.	490	427	5
	Short Trans.	476	407	2
Over 102 to 127, incl	Longitudinal	483	427	8
	Long-Trans.	490	427	5
	Short Trans.	469	400	2
Over 127 to 152, incl	Longitudinal	483	427	7
	Long-Trans.	483	421	4
	Short Trans.	469	400	2
Over 152 to 178, incl	Longitudinal	476	427	6
	Long-Trans.	476	414	3
	Short Trans.	462	393	2
Over 178 to 203, incl	Longitudinal	469	421	5
	Long-Trans.	476	414	3
	Short Trans.	455	393	2
Over 203 to 229, incl	Longitudinal	469	421	5
	Long-Trans.	469	407	3
	Short Trans.	455	386	2

3.3.2 Corrosion-Resistance

Resistance to stress-corrosion cracking and to exfoliation-corrosion shall be considered acceptable if the plate conforms to the requirements of 3.3.2.1 and 3.3.2.2.

3.3.2.1 Electrical Conductivity

Shall be not lower than 38.0% IACS (International Annealed Copper Standard) (22.6 MS/m), determined on the surface of the tensile coupon.

3.3.2.2 Stress-Corrosion Susceptibility Factor (SCF)

Shall be not greater than 32.0 (220), determined by subtracting the electrical conductivity, AA.A% IACS (12 times BB.B MS/m) from the long-transverse yield strength, XX.X ksi (YYY MPa).

3.3.2.2.1 Example

For plate 4.0 inches (102 mm) nominal thickness:

TABLE 3A

Inch/Pound Units		
33.5	unacceptable	73.1 ksi - 39.6% IACS
28.6	acceptable	68.8 ksi - 40.2% IACS

TABLE 3B

SI Units		
MS/m = 228	unacceptable	504 MPa - 12 × 23
MS/m = 194	acceptable	474 MPa - 12 × 23.3

3.3.2.3 Plate not meeting the requirements of 3.3.2.1 shall be given additional precipitation heat treatment or reheat treated. After such treatment, if all specified properties are met, the plate is acceptable.

3.3.3 Exfoliation-Corrosion Test

Plate shall not exhibit exfoliation-corrosion greater than that illustrated by Photo B, Figure 2, of ASTM G 34.

3.3.4 Stress-Corrosion Test

Specimens shall show no evidence of stress-corrosion cracking when stressed in the short-transverse direction at 35.0 ksi (241 MPa) for 20 days in accordance with ASTM G 47.

3.3.5 Fracture Toughness

When specified, plate shall meet the values for K_{Ic} specified in Table 4. For T-L and L-T test directions on plate over 3 to 4 inches (76 to 102 mm), inclusive, in nominal thickness, use specimens 2-inch (51-mm) minimum thickness centered at T/2; and for plate over 4 inches (102 mm) in nominal thickness, use specimens 2-inch (51-mm) minimum thickness centered at T/4. For the S-L test direction, the test specimen shall be centered at T/2. Required specimen orientation(s) shall be specified by purchaser.

TABLE 4A - MINIMUM FRACTURE TOUGHNESS PARAMETERS, INCH/POUND UNITS

Nominal Thickness, Inches	L-T ksi√inch	T-L ksi√inch	S-L ksi√inch
3.0 to 4.0, incl	31	26	24
Over 4.0 to 5.0, incl	30	25	24
Over 5.0 to 6.0, incl	29	23	24
Over 6.0 to 7.0, incl	27	22	23
Over 7.0 to 8.0, incl	26	22	23
Over 8.0 to 9.0, incl	26	21	22

TABLE 4B - MINIMUM FRACTURE TOUGHNESS PARAMETERS, SI UNITS

Nominal Thickness, Millimeters	L-T MPa√m	T-L MPa√m	S-L MPa√m
76 to 102, incl	34	28	26
Over 102 to 127, incl	33	27	26
Over 127 to 152, incl	32	25	26
Over 152 to 178, incl	30	24	25
Over 178 to 203, incl	28	24	25
Over 203 to 229, incl	28	23	24

3.3.6 Fatigue Resistance

When specified, 4 to 9-inch (102 to 229-mm) thick plate shall meet the values for fatigue life specified in Table 5. Two fatigue specimens from each end of the plate shall be sampled in the long transverse grain direction. These specimens are to be removed from the T/2, W/2 location. Fatigue testing shall be conducted in air at 70 °F ± 5 (21 °C ± 3) in accordance with ASTM E 466. They are to be tested at an R-ratio of 0.1, at a maximum stress of 35.0 ksi (241 MPa) and shall meet the fatigue requirements shown in Table 5. When agreed upon by the supplier and purchaser, alternate specimen location and test conditions, and fatigue life requirements may be used.

TABLE 5 - FATIGUE LIFE REQUIREMENTS

Minimum cycles per test	90,000 cycles
Average of 4 tests	120,000 cycles
Runout	300,000 cycles

3.4 Quality

Plate, as received by purchaser, shall be uniform in quality and condition, sound, and free from foreign materials and from imperfections detrimental to usage of the plate.

3.4.1 Each plate shall be ultrasonically inspected in accordance with ASTM B 594 and shall meet the requirements of 3.4.1.1 or 3.4.1.2 as applicable.

3.4.1.1 Plates weighing 2000 pounds (907 kg) and under shall meet the requirements for ultrasonic class A.

3.4.1.2 The ultrasonic class for plates weighing over 2000 pounds (907 kg) shall be as agreed upon by purchaser and vendor.

3.5 Tolerances

Shall conform to all applicable requirements of ANSI H35.2 or ANSI H35.2M.